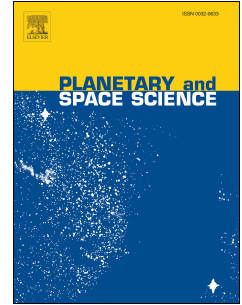




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1 Calibration of NOMAD on ExoMars Trace Gas Orbiter: Part 3 - LNO
2 validation and instrument stability

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16 1. Introduction

17 The ExoMars program consists of two missions designed to study the trace gases of the mar-
18 tian atmosphere but also to acquire information on potential ongoing geological and biological
19 processes on the surface of Mars (Vago et al., 2015). Since April 2018, the four instruments aboard
20 the ESA/Roscosmos ExoMars Trace Gas Orbiter mission has acquired observations of both the

21 atmosphere and surface of Mars. Among them the NOMAD instrument (Nadir and Occultation
22 for MArS Discover), led by the Belgian Institute for Space Aeronomy (BIRA-IASB), is a suite of
23 three spectrometers spanning the UV and IR spectral range: SO (Solar occultation), LNO (limb,
24 nadir, and occultation) and UVIS (ultraviolet-visible). The three channels work separately but
25 are all controlled via a single main electronic interface (Neefs et al., 2015). The two first channels
26 are infrared spectrometers based upon the SOIR (Solar Occultation in the InfraRed) instrument
27 aboard the Venus Express mission (Nevejans et al., 2006).

28 The LNO channel is a compact high-resolution echelle grating spectrometer with an acousto-
29 optic tunable filter (AOTF) working in the infrared domain from $2.3 \mu\text{m}$ to $3.8 \mu\text{m}$ ($4250\text{-}2630$
30 cm^{-1}) with a resolving power ($\lambda/\Delta\lambda$) of around 10000, specially designed for nadir observation.
31 With such high resolving power combined with the near-circular orbit of TGO permitting 12 orbits
32 in one sol, promoting a global coverage of the planet, the NOMAD-LNO instrument is perfectly
33 suited to study the martian surface and atmosphere.

34 The main objective of this article is to propose an original calibration procedure, adaptable
35 for the full dataset of NOMAD-LNO. This calibration is complementary to the one proposed
36 by Thomas et al. (2021) who developed a fully empirical method using in-flight data. In their
37 paper the LNO ground calibration, occultation and nadir boresight pointing vectors, detector
38 characterisation and illumination pattern are covered. A combination of several observation of the
39 sun is used to derive instrument temperature effects such as the shape and intensity of a LNO
40 spectrum. The radiometric calibration is done by assuming temporal stability of the instrument
41 and directly using solar observation to calibrate nadir observation.

42 In this paper we will not assume temporal stability of the instrument. Our approach is thus
43 able to investigate the temporal evolution of the instrumental sensitivity, which is expected to vary
44 due to degradation by energetic particles. This approach will be based on an empirical continuum
45 removal to take into account the departure between actual blaze function and its theoretical form.
46 By construction, our approach is thus more robust but may fail to model some instrumental
47 effect such as the temperature dependence of the blaze and AOTF transfer function on the raw
48 continuum of an LNO spectrum. The main calibration of NOMAD-LNO is well described in

49 Thomas et al. (2021). This complementary work aims to validate the calibration of LNO but also
50 to give additional information about instrumental transfer function and instrumental line shape.

51 **2. NOMAD LNO instrument**

52 The optical design of the LNO spectrometer is identical to that of SO and therefore very similar
53 to SOIR (Nevejans et al., 2006; Vandaele et al., 2013), it is a combination of a high-dispersion
54 echelle grating along with an AOTF and a cooled detector. The main advantage of using an echelle
55 grating is that the full height of the detector can be used to register spectral lines (Neefs et al.,
56 2015), which greatly improves the SNR after column binning. In the spatial direction, the detector
57 contains 256 rows of pixels and only 144 of the 150 illuminated spatial lines are read out (Thomas
58 et al., 2021). The AOTF is placed as diffraction order sorting devices in front of the spectrometer
59 section to avoid order overlap at the output of the grating (Neefs et al., 2015). More information
60 regarding the LNO channel, its working principle, and technical details can be found in previous
61 articles (Neefs et al., 2015; Thomas et al., 2016, 2021).

62 Before starting a nadir observation, the detector needs to be cooled to its operating temperature
63 of approximately 90 K (Neefs et al., 2015) to ensure that the thermal noise of the environment
64 is not measured by the spectrometer. The AOTF will select the diffraction order to be observed,
65 therefore, three types of observations are possible :

- 66 • Miniscan: Only a fraction of the entire spectral range is measured. The AOTF frequency
67 slowly varies to observe the signal across a few diffraction orders and the transition between
68 them.
- 69 • Fullscan: The entire spectral range of the channel is measured. The AOTF input varies by
70 large steps, covering one diffraction order at a time.
- 71 • Dedicated scan: special AOTF frequencies are observed to select the corresponding diffraction
72 order (for instance only the one where methane absorption bands are present).

73 As well as the SO channel, LNO can operate looking toward the sun which allows measurement
74 of the solar lines for precise in-flight calibration. The solar miniscans and fullscans are precious

75 observation that provides complementary information on the instruments. Solar miniscans allow
 76 a precise determination of the AOTF transfer function shape and the Instrument Line Shape but
 77 each miniscan covers only a limited spectral range. A first in-flight calibration using miniscans
 78 has been proposed by Liuzzi et al. (2019). Fullscans are dedicated to spectral and photometric
 79 calibration as the whole spectral range is covered. In this work, the calibration model built for
 80 LNO is based on the solar fullscans observations.

81 **3. Data**

82 *3.1. Solar fullscan*

83 The NOMAD-LNO fullscans are solar observation made for calibration purposes. The instru-
 84 ment, normally in nadir position, is pointing toward the sun. The choice of using solar fullscans
 85 was made for two reasons. First, there are not enough miniscans to cover all diffraction orders with
 86 a significant amount of data while fullscans always cover the whole spectral range which allows
 87 testing the time dependence of the calibration. Second, it is important to estimate the instru-
 88 mental sensitivity over the whole diffraction order range. As of June 2020, six solar calibrations
 89 have been performed on the following dates: 02/07/2018, 01/11/2018, 14/03/2019, 09/06/2019,
 90 07/12/2019 and 05/01/2020. A typical fullscan observation is shown in figure 1. The x-axis is the
 91 pixel number (sometimes also called spectel), the y-axis is the diffraction order (i.e. the AOTF
 92 frequency) and the contours shows the sensitivity of the detector to incoming solar radiation (in
 93 ADU for Analog to Digital Units). As a function of the pixel number, the sensitivity is best be-
 94 tween 100 and 250 and very low between 1 and 100. This shape is related to the Blaze function
 95 as it reaches lower values on the edge of the detector (pixel number 1 and 320). As a function of
 96 the diffraction order, the sensitivity is good around order 130 and best between order 160 and 200.
 97 This is related to the scientific goal of NOMAD, as it was designed to study the atmosphere (CO,
 98 CO₂, and H₂O between orders 163 and 192) and its traces gases (CH₄ between orders 130-136).
 99 Raw data (in ADU) are already dark subtracted directly onboard. Before using the fullscans as
 100 raw data for the calibration, a normalization needs to be done to account for the integration time
 101 t_{int} , the spectral resolution Δ_ν (varying across the spectral range), the number of accumulation
 102 NOA of recorded spectra, and the binning bin done before transmitting the data to Earth. The

103 normalization equation follows :

$$S_{obs} = \frac{Counts(ADU)}{t_{int} \cdot \Delta\nu \cdot NOA \cdot bin} \quad (1)$$

104 Where Counts is the dark-subtracted raw spectrum across the 320 spectels. For the LNO
105 fullscan, instrument parameters are:

- 106 • $t_{int} = 2$ ms
- 107 • $0.1469 \leq \Delta\nu \leq 0.2649$ cm^{-1} between order 110 and 215 (Liuzzi et al., 2019)
- 108 • $NOA = 78$
- 109 • $bin = 24$ as the 24 central lines are illuminated by the sun and stacked to increase the signal
110 for the calibration

111 These parameters are constant for all LNO fullscans. The spectrum S_{obs} is considered the actual
112 observation of the sun. Each sequence of a fullscan starts with the order 110 and ends with the
113 order 215. This sequence is then repeated several times (between 14 and 34 times). Due to the
114 very weak signal-to-noise ratio on the edge of the detector (spectels 1 to 50) only spectels 50 to
115 320 are considered for the calibration.

116 3.2. Construction of a reference synthetic solar spectrum for NOMAD

117 The sun is the best blackbody available for in-flight calibration because its spectrum is well
118 defined and its time variations are minimal. The synthetic solar spectrum that we use to compare
119 LNO solar observations is derived from the Fourier Atmospheric Chemistry Experiment (ACE)
120 spectrometer data aboard SciSat-1, operating in the spectral range 2-13 μm ($750 - 4430$ cm^{-1}) at
121 a very high spectral resolution (0.02 cm^{-1}) (Hase et al., 2010). The processed data available is
122 a transmittance solar spectrum (called S_T hereafter) which was converted to spectral irradiance
123 spectrum (in $W.cm^{-2}.(cm^{-1})^{-1}$) with a fit of the spectral continuum of the *NewKurucz* (Chance
124 and Kurucz, 2010) solar spectrum S_{Kurucz} from the MODTRAN spectral database. To do so we
125 first defined the continuum by a convolution with a median Gaussian filter $G(0)$ with a large
126 standard deviation (σ) to avoid the contribution of solar lines:

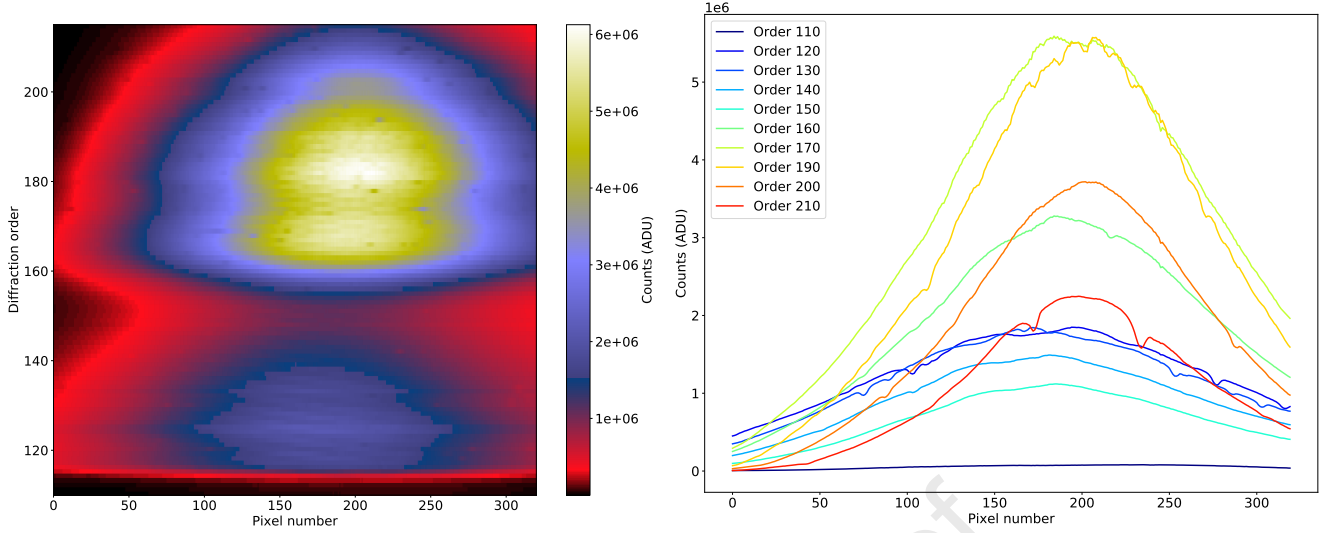


Figure 1: Example of a typical NOMAD-LNO solar fullscan (observation of 14/03/2019, sequence number 10 out of 15). Here the 24 central lines illuminated are summed to promote a better SNR. The left figure shows the complete fullscan across the whole LNO spectral range (order 110 to 215). Each line of the image represents a spectrum at a particular diffraction order sampled on the 320 spectral spectels of the detector (i.e the wavenumbers). The colors represent the intensity (expressed in ADU) measured by the detector and solar lines are visible as spots. The right figure shows a cross-section within the fullscan at particular diffraction orders. Here the characteristic “bell” shape of the continuum is clearly visible, the width and height of each spectrum is order-dependent and are related to the instrument-specific design (AOTF + blazed grating). Weak solar lines are also visible which illustrates the high spectral resolution of the instrument.

$$S_{continuum} = S_{Kurucz} \star \mathcal{G}(0, \sigma) \quad (2)$$

127 Then, the solar continuum $S_{continuum}$ was interpolated in the wavenumbers sampling of the
 128 ACE spectrum to maintain the very high spectral resolution. The spectrum is then converted to
 129 spectral radiance (in $W.cm^{-2}.(cm^{-1})^{-1}.sr^{-1}$) by dividing the interpolated $S_{continuum}$ spectrum by
 130 the solid angle (in sr units) with which the sun is observed by LNO. In this case the sun is entirely
 131 resolved by the detector therefore the solid angle is simply the surface of the sun divided by the
 132 square of the Mars-Sun distance (in km). Here we considered the mean Mars-Sun distance (1.524
 133 AU). We also integrate the Mars-Sun distance ($d_{\odot\sigma}$ (in AU) in $S_{continuum}$ since the *NewKurucz*
 134 solar spectrum was measured from Earth orbit at a distance of 1 AU. This leads to the final
 135 spectrum S_L :

$$S_L = S_{continuum} \cdot S_T / d_{\odot\sigma}^2 \quad (3)$$

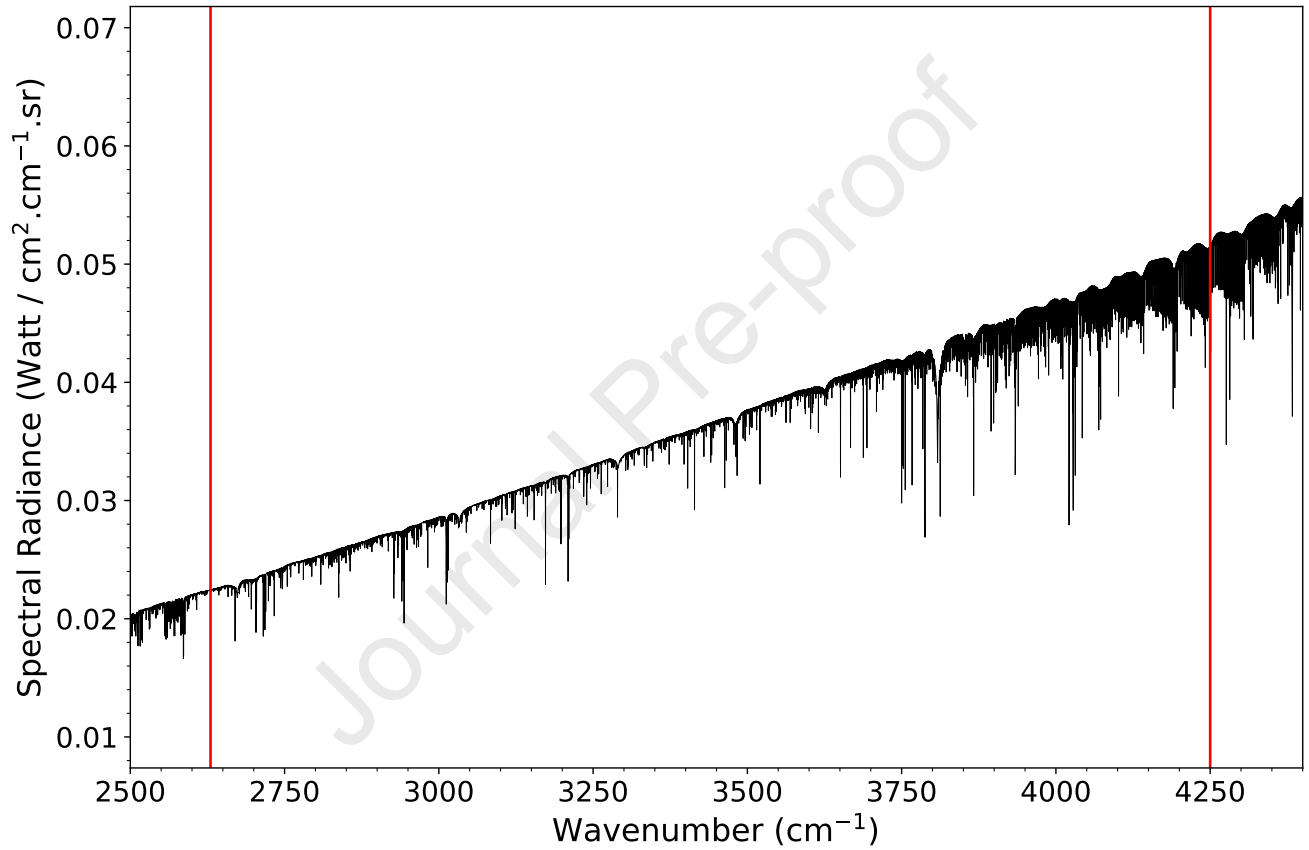


Figure 2: Solar spectrum (S_L) used in this study in spectral radiance and the mean distance Mars-Sun (1.524 AU). The very high spectral resolution of the Fourier ACE spectrometer allows the identification of weak solar lines, which is necessary for the calibration of LNO. The LNO spectral range is shown as red vertical lines.

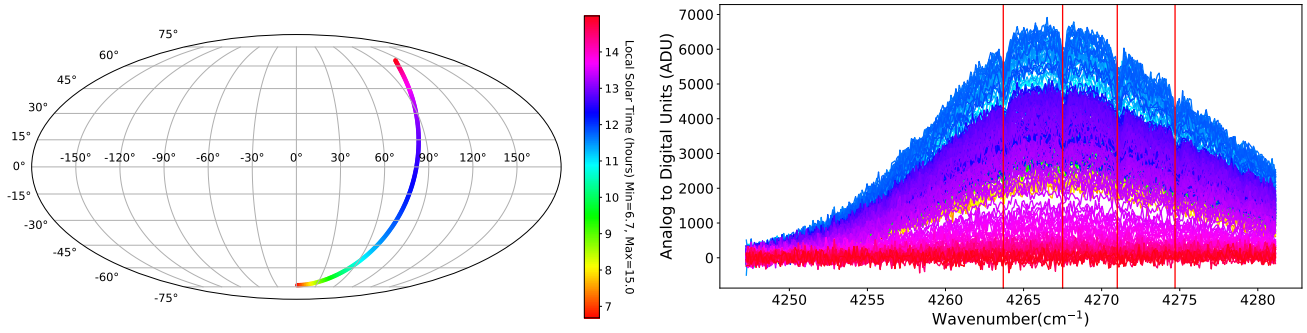


Figure 3: Order 189: groundtrack (left) and raw spectra (right) of 18/09/2018, the colors refer to local solar time. The low-altitude near-circular orbit of TGO allows observations at various local times. The characteristic 'bell' shape of LNO spectra is visible, especially when the signal is strong at noon local solar time (in blue). The atmospheric bands are also visible, here red vertical lines indicate the CO absorption bands.

136 3.3. NOMAD-LNO observation (*nadir*)

137 The science phase of the LNO channel started in March 2018. During one orbit (*nadir*-looking),
 138 one to six diffraction orders can be measured, for instance when 3 orders are measured the ob-
 139 servation cycle lasts 15 seconds and is decomposed into 3 periods of 5 seconds corresponding to
 140 a different configuration of the AOTF (Vandaele et al., 2015, 2018). Thus, for the same cycle,
 141 three diffraction orders can be measured, each measurement is completed by a measurement of the
 142 "Dark" which is then subtracted directly on board (Vandaele et al., 2018). In April 2021, more
 143 than 12,810 files are available, each file corresponds to a particular diffraction order during one
 144 orbit. Not all diffraction orders are measured in each orbit and most of the time a combination of
 145 useful orders for a science objective are acquired together, such as orders 131-136 (CH_4), 146-150
 146 (CO_2), 167-171 (H_2O), and 186-192 (CO) (Liuzzi et al., 2019; Vandaele et al., 2019). We decide
 147 here to illustrate our calibration with order 189 due to the significant amount of data and for its
 148 scientific interest (Smith, 2021). Thus, for a particular order, the time necessary to obtain a global
 149 map coverage of the planet can be spread over a long time, typically one martian year. To exclude
 150 any temporal variation in the instrument that could bias the calibration and the reflectance map
 151 produced, it is important to test the time dependence of the instrument on the photometric sensi-
 152 tivity to look at the potential aging of the detector before calibrating data spanning several years.
 153 A typical LNO orbit is shown in figure 3 for order 189.

154 4. Method

155 The calibration aim was to build a model to estimate the spectral conversion (wavenumbers
 156 in cm^{-1} for each spectels of the detector) and the photometric sensitivity (conversion factor from
 157 ADU to spectral radiance). The model must be versatile enough to face the uncertainties of some
 158 instrumental functions, such as the AOTF transfer function and the grating blaze function.

159 The method used here is based on the usual comparison between a real solar observation and
 160 a simulated solar spectrum. The observations are LNO solar fullscans and we aim to simulate
 161 the same observation by accounting for the AOTF and grating effects. The comparison is done
 162 after spectra continuum correction to work out the baseline uncertainties. This approach is of
 163 particular importance for correct calibration of nadir data, for which there is no observation free
 164 of atmospheric absorption (Liuzzi et al., 2019). The calibration steps we perform are the following
 165 :

- 166 1. First-order spectral calibration: conversion of detector spectel number and diffraction order
 167 to wavenumber (Section 4.1)
- 168 2. Calibration inversion, by minimizing the residual between observation and simulation, fol-
 169 lowing these steps:
 - 170 (a) Application of the grating blaze function and AOTF transfer function shape (Section
 171 4.2.1)
 - 172 (b) Simulation of the Instrument Line Shape (ILS) (Section 4.2.2)
 - 173 (c) Integration of a residual wavenumber shift (Section 4.2.3)
 - 174 (d) Continuum removal to remove the shape imposed by the instrumental functions (Section
 175 4.2.4)
 - 176 (e) Normalization and estimation of the sensitivity (Section 4.2.5)
 - 177 (f) Computation of the residue (Section 4.3)
- 178 3. By construction, the result of the minimization leads to the estimation of the instrumental
 179 sensitivity

180 An illustration of the calibration pipeline with the different corresponding sections is given figure
 181 4.

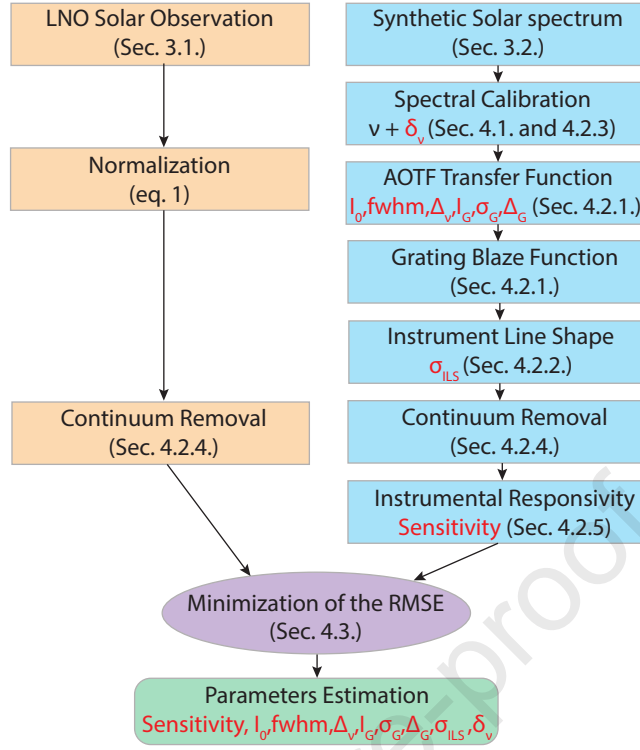


Figure 4: Flowchart summarizing the steps of the calibration model. The parameters estimated by the inversion are indicated in red.

182 4.1. Spectral calibration

183 The very first step to perform is to compute the spectral calibration which is based on the
 184 position of well-defined solar lines in the calibration spectra that can be used to match a line's
 185 frequency with the spectral number in which it falls (Liuzzi et al., 2019). The diffraction order
 186 number is related to the AOTF frequency and the aim is to get a relationship between the spectral
 187 number of the detector and the wavenumber, at a corresponding diffraction order. This relationship
 188 is modeled by a second-order polynomial :

$$\frac{\nu}{m} = F_0 + F_1.p + F_2.p^2 \quad (4)$$

189 Where ν is the wavenumber, m is the diffraction order and p is the spectral number (from 1
 190 to 320). The coefficients F_0 , F_1 and F_2 were obtained by Liuzzi et al. (2019) using Miniscan
 191 acquisitions and a total of 70 solar lines. The central spectral wavenumber of the considered order
 192 is noted ν_0 for spectral 160. The Free Spectral Range (FSR) is the wavelength range in which
 193 there is no overlap by adjacent orders. It can be computed using equation 4 at the spectral 160

194 (center of the detector) and gives a value of 22.545 cm^{-1} which is very close to what is reported in
 195 Neefs et al. (2015) with a value of 22.56 cm^{-1} . The FSR is expected to be a constant for echelle
 196 gratings (Neefs et al., 2015). To take into account small residual wavenumber shift, we introduced
 197 a parametrization described in section 4.2.3.

198 4.2. Photometric calibration

199 To simulate the instrumental effects, we need to take into account the instrumental transfer
 200 function T (here due to the grating and the AOTF), and the instrumental line shape ILS that
 201 blurred the signal in spectral domain. A simplified expression of the instrumental effects is:

$$S_{sim}(\nu) = (S_L(\nu).T(\nu)) \star ILS \quad (5)$$

202 Where $S_L(\nu)$ is the simulated solar spectrum described in section 3.2 and $T(\nu)$ is the instru-
 203 mental transfer functions described hereafter. In the next sections, we will describe precisely the
 204 instrument model taken into account. We will emphasize for all functions the main parameter
 205 ν (representing the wavenumber) and the unknown parameters that our inversion scheme will
 206 estimate.

207 4.2.1. Instrumental transfer function

208 *Grating blaze function.* A diffraction grating or echelle spectrometer is an optical device which
 209 allows diffraction of the incident light into different diffraction orders. Slit spacing controls the
 210 extent of the diffraction for a given wavelength. For the LNO spectrometer, a “blazed grating” is
 211 used where the grooves are tilted in a special angle, called the “Blaze angle”. This angle is generally
 212 chosen to optimize the efficiency of the diffraction for a given wavelength. The commonly adopted
 213 geometry is called “Littrow configuration” in which the Blaze angle is chosen so that the incident
 214 and diffraction angle are the same, and are equal to the Blaze angle. Moreover, the LNO has
 215 a near-Littrow configuration (Neefs et al., 2015) to slightly separate the incident beam from the
 216 diffracted beam. Thus, the grating is tilted so that both beams make a small γ angle between the
 217 perpendicular plane to the grating surface and the grooves. It is also necessary to take into account
 218 a slight deviation angle i of the incident angle with respect to the Blaze angle in the perpendicular
 219 plane to the grooves. The grating equation becomes (Neefs et al., 2015):

Parameters	Value	Precision	Unit
Blaze angle θ_B	63.43	± 0.1	$^\circ$
Groove density σ (at 0°C)	4.033512		lines/mm
Number of grooves N	≥ 765		
Groove spacing d	248.06	± 0.001	μm
Off-Littrow angle γ	2.75	± 0.0025	$^\circ$
Off-Blaze angle i	0	± 0.016	$^\circ$

Table 1: Grating characteristics of the LNO channel, from Neefs et al. (2015).

$$\frac{m}{\nu} = \frac{(\cos \alpha + \sin \beta) \cdot \cos i \cdot \cos \gamma}{\sigma} = \frac{2 \sin \theta_B \cdot \cos i \cdot \cos \gamma}{\sigma} \quad (6)$$

220 Where m is the diffraction order, α is the incident angle, β is the diffraction angle, θ_B is the
 221 Blaze angle, σ is the grating groove density (lines/mm), and ν is the wavenumber. The blaze
 222 grating characteristics of the LNO channel are summarized in table 1. The grating efficiency is
 223 estimated at 85% in the center of the grating and falls to 40% at the edges (Neefs et al., 2015). For
 224 each wavenumber within each order, we compute the diffraction angle β using equation 6. Then,
 225 we estimate the blazing efficiency for which we adopt the following relation :

$$T_{Blaze}(\nu) = C \cdot \left[\text{sinc} \left(\nu \cdot \frac{\sigma \cdot \cos \gamma \cdot \cos \alpha}{\cos \alpha_B} \cdot (\sin \alpha_B + \sin(\beta - \theta_B)) \right) \right]^2 \quad (7)$$

226 Where $\alpha_B = \theta_B - \alpha$, the angle between the incident beam and the normal of the facet in
 227 the perpendicular plane to the grooves, and the grating geometric parameters $C = 1$ if $\alpha > \beta$
 228 or $\left(\frac{\cos \beta}{\cos \alpha}\right)^2$ if $\alpha < \beta$. The measured global shape of NOMAD spectra is partly determined by this
 229 $T_{Blaze}(\nu)$ function.

230 The LNO grating is made of an aluminum alloy (Neefs et al., 2015) and is therefore subject to
 231 compressions/expansions due to temperature changes. This would lead to groove density variations
 232 and shift the position of the function on the detector spectels. If the temperature within the
 233 spectrometer is known, then this effect can be taken into account considering the following formula
 234 describing the groove length σ . From a known groove length σ_1 at temperature T_1 it is possible to
 235 estimate the groove length σ_2 at temperature T_2 using the thermal expansion equation (Marquardt
 236 et al., 2000):

a	$-4.1272 \cdot 10^{-3}$
b	$-3.0640 \cdot 10^{-6}$
c	$8.7960 \cdot 10^{-8}$
d	$1.0055 \cdot 10^{-10}$
T_1	$24.5^\circ C$
σ_1	$248.06 \mu m$

Table 2: Reference groove spacing and temperature for the LNO channel from Marquardt et al. (2000).

$$\sigma_2 = \sigma_1 \cdot \frac{(1 + (a + b.T_2 + c.T_2^2 + d.T_2^3))}{(1 + (a + b.T_1 + c.T_1^2 + d.T_1^3))} \quad (8)$$

237 Where a , b , c and d are instrument-specific coefficients for the LNO channels, T_2 is the given
 238 instrument temperature, T_1 is the design temperature and σ_1 is the reference groove spacing at
 239 design temperature (see table 1). Table 2 summarizes their value. Because the blaze function
 240 mainly controls the position of a NOMAD raw spectrum on the 320 pixels, especially the position
 241 of maximum intensity, and since the temperature of the grating is not perfectly known, our strategy
 242 was to perform a rough estimation: we average all data from the same fullscan for each order and
 243 perform a regression to estimate the temperature to use for each order. The residual discrepancies
 244 between blaze function and real data are removed by continuum removal.

245 *AOTF transfer function shape.* The AOTF is an electro-optical device based on the Bragg diffrac-
 246 tion, the interaction between acoustic and electromagnetic waves, in a birefringent crystal of Tel-
 247 lurium dioxide (TeO_2). Under the application of a radio-frequency (using a Transducer) the crystal
 248 vibrates and its internal structure is alternatively compressed and dilated through the crossing of
 249 the oscillating acoustic wavefront. This results in a periodic modification of its optical properties,
 250 its refractive index, thus producing diffraction of the incident light. The intensity of the diffraction
 251 is proportional to the amplitude of the acoustic signal applied to the crystal. In addition, changing
 252 the acoustic frequency applied to the crystal alters the periodicity of the refractive index variation
 253 and therefore the wavelength at which the light is diffracted. The AOTF crystal acts mainly as
 254 a filter for selecting the desired order of diffraction which will be transmitted to the diffraction
 255 grating. Thus, if no radiofrequency is applied to the crystal (AOTF off) then the spectrometer
 256 measures the thermal noise and the background noise also called “dark observation” which are used

257 to correct the specific noises of the detector environment. The main characteristics of the AOTF
 258 used for the LNO channel are well described in Neefs et al. (2015). The AOTF transfer function
 259 shape is usually modeled by a sinc-square function. However, several tests showed that this model
 260 does not accurately represent the NOMAD-LNO AOTF (Liuzzi et al., 2019). This may be due to
 261 the contributions from adjacent orders which are not negligible and can be related to two main
 262 reasons :

- 263 • The AOTF input radiofrequency is not centered on the AOTF transfer function with respect
 264 to the observed order, so the AOTF admits throughput of adjacent orders.
- 265 • The Free Spectral Range of the grating is smaller than the detector's sensitive area causing
 266 invasion from adjacent orders.

267 To properly account for this effect, we tested several models that would increase the contribution
 268 from adjacent orders in the AOTF transfer function shape (using multiple sinc-square, using one
 269 sinc-square per order or adding a linear function to add a background contribution). The most
 270 promising model turns out to be the use of a Gaussian function along with the usual sinc-square
 271 function which is similar to what is reported in Liuzzi et al. (2019). The AOTF transfer function
 272 shape follows :

$$T_{sinc}(\nu, I_0, fwhm, \Delta_\nu) = I_0 \cdot \left[\text{sinc} \left(0.886 \cdot \frac{\nu - \nu_0 - \Delta_\nu}{fwhm} \right) \right]^2 \quad (9)$$

$$T_{gauss}(\nu, I_G, \sigma_G, \Delta_G) = I_G \cdot \exp \left[-\frac{(\nu - \nu_0 - \Delta_G)^2}{\sigma_G^2} \right] \quad (10)$$

$$T_{AOTF}(\nu, I_0, fwhm, \Delta_\nu, I_G, \sigma_G, \Delta_G) = T_{sinc} + T_{gauss} \quad (11)$$

273 Where $fwhm$ is the full width at half maximum of the sinc-square function, Δ_ν is the sinc-square
 274 shift in wavenumber from the central wavenumber of the considered order ν_0 , I_0 the sinc-square
 275 amplitude, I_G the Gaussian amplitude, σ_G the Gaussian standard deviation and Δ_G the shift in
 276 wavenumber of the Gaussian function. These are the six free parameters to estimate related to
 277 the AOTF. Adding a Gaussian function lowers the contribution of the transfer function side lobes,

278 which were related to the sinc-square function, but widens the shape of the main lobe and prevents
 279 the function to reach zero intensity between the main lobe and the first side lobe. This allows the
 280 increase of the contribution from the first nearby order. The main lobe is still centered on the
 281 central order which guarantees that most of the filtered signal is related to this order.

282 *Final expression.* The simulated solar spectrum is therefore constrained by the shape and intensity
 283 of the AOTF transfer function and the grating blaze function. The contribution from one order
 284 depends on the product of both functions at one particular wavenumber (i.e., on a spectral number).
 285 Contribution from adjacent orders is summed on the 320 spectral detector spectels which samples
 286 the wavenumber of the central order. The expression of a simulated solar observation by LNO is
 287 shown in equation 12.

$$S_{sim}^{instru}(\nu, I_0, fwhm, \Delta\nu, I_G, \sigma_G, \Delta G) = \sum_{m-\Delta m}^{m+\Delta m} S_L \cdot (T_{Blaze} \cdot T_{AOTF}) \quad (12)$$

288 The calibration proposed here was made considering 3 adjacent orders ($\Delta m = 3$) following the
 289 work of Liuzzi et al. (2019) who has shown that using 3 adjacent orders amounts to considering
 290 more than 99% of the total flux for the LNO channel. The absolute amplitude of the blaze and
 291 AOTF transmittance are estimated jointly as a general instrument sensitivity (see section 4.2.5).
 292 T_{Blaze} and T_{AOTF} are usually normalized.

293 4.2.2. Instrument Line Shape (ILS)

294 The simulated solar spectrum (S_{sim}^{instru}) is first convolved to the Instrumental Line Shape (ILS),
 295 modeled here by a centered Gaussian function $\mathcal{G}(0, \sigma_\nu)$ defined by a standard deviation (σ_ν) similar
 296 to what is reported in Liuzzi et al., 2019 (Liuzzi et al., 2019). This result in a smoothed spectrum
 297 S_{sim}^{smooth} defined as:

$$S_{sim}^{smooth}(\nu, \sigma_\nu) = S_{sim}^{instru}(\nu) \star \mathcal{G}(0, \sigma_\nu) \quad (13)$$

298 This step is crucial to simulated realistic solar lines shape and to characterize the actual
 299 wavenumber resolution of the instrument. Also, as the initial solar spectrum (S_L) used for the cal-
 300 ibration has a much higher spectral resolution than NOMAD, the convolution allows the simulated

301 spectrum to match the theoretical spectral resolution of the observation. To compute the spectral
 302 resolution as a function of the diffraction order, we first compute the AOTF driver frequency to
 303 AOTF central wavenumber following the tuning relation :

$$\nu_0 = G_0 + A.G_1 + A^2.G_2 \quad (14)$$

304 where ν_0 is the central wavenumber and A is the AOTF frequency. The coefficients G_0 , G_1 ,
 305 and G_2 were determined by Liuzzi et al. (2019) using miniscans. Then, we compute the spectral
 306 resolution at the desired diffraction order :

$$sr = sr_0 + \nu_0.sr_1 + \nu_0^2.sr_2 \quad (15)$$

307 The coefficients sr_0 , sr_1 and sr_2 are empirical values obtained by Liuzzi et al. (2019) using
 308 miniscans. The spectral resolution is order-dependent and is estimated with the central wavenum-
 309 ber (ν_0) of each order. This value is used as initialization and we treat σ_ν as a free parameter.

310 4.2.3. Residual wavenumber shift

311 The spectral calibration (see section 4.1) is not perfectly constant over time and solar lines
 312 are slightly shifted from one observation to another. Despite a first rough estimation of the
 313 temperature of the grating blaze function some residual shift may persist due to small variation of
 314 the blaze function within the same order. Since the temperature of the grating is not known such
 315 an effect must be taken into account for precise calibration. Our approach is to estimate the shift
 316 for each observation to possibly improve the formulation of eq. 4. We introduce a residual shift in
 317 wavenumber (δ_ν) between the observation (S_{obs}) and simulation (S_{sim}^{smooth}), the simulated spectra
 318 was resampled on the observed wavenumber with the linear resampling method:

$$S_{sim}^{smooth,resamp}(\nu) = S_{sim}^{smooth}(\nu + \delta_\nu) \quad (16)$$

319 with the wavenumber shift (δ_ν) set as a free parameter.

320 4.2.4. Continuum removal

321 The justification of this continuum removal step is done section 5. We propose to use the
 322 continuum estimation noted \mathfrak{C} , using the following approach :

$$S_{sim}^{smooth,resamp,flat}(\nu) = \frac{S_{sim}^{smooth,resamp}(\nu)}{\mathfrak{C}(S_{sim}^{smooth,resamp}(\nu))} \cdot \overline{S_L} \quad (17)$$

$$S_{obs}^{flat}(\nu) = \frac{S_{obs}(\nu)}{\mathfrak{C}(S_{obs}(\nu))} \cdot \overline{(S_{obs})} \quad (18)$$

323 Where $\overline{S_L}$ is the mean solar radiance taken from the input solar spectrum at the desired order
 324 and $\overline{(S_{obs})}$ is the mean of the normalized observation recorded by the detector for the desired order
 325 and sequence. Both means are computed with spectels 50 to 320. We used the function \mathfrak{C} following
 326 the Asymmetric Least Squares smoothing (Eilers et al., 2005) with the following parameters:

- 327 • Smoothness $\lambda = 1.10^2$;
- 328 • Asymmetry $p = 0.99$;
- 329 • Number of iterations : 10 ;

330 4.2.5. Sensitivity

331 The sensitivity of the instrument is computed using the following equation :

$$Sensitivity = \frac{\left\| \overline{S_{sim}^{smooth,resamp,flat}}(W.cm^{-2}.(cm^{-1})^{-1}.sr^{-1}) \right\|}{\left\| S_{obs}^{flat}(ADU.(cm^{-1})^{-1}.s^{-1}) \right\|} \quad (19)$$

332 Where $\overline{S_{obs}^{flat}}$ is the mean of the normalized observation after continuum correction as stated in
 333 equation 18 and $\overline{S_{sim}^{smooth,resamp,flat}}$ the mean of simulated spectra after the continuum correction.
 334 The two means are estimated for spectels 50 to 320. We choose to take the mean instead of the
 335 intensity at the center of the detector because it is a more robust indicator than the value of a
 336 single spectel which is subject to noise. The sensitivity simply expresses the mean radiance per
 337 counts per second of the 270 considered spectels. Once the sensitivity is known, the observation
 338 data can be calibrated into spectral radiance following:

$$\overline{S_{obs}^{flat,cal}}(W.cm^{-2}.(cm^{-1})^{-1}.sr^{-1}) = Sensitivity.S_{obs}^{flat}(ADU.(cm^{-1})^{-1}.s^{-1}) \quad (20)$$

339 With $\overline{S_{obs}^{flat}}$ the normalized observation after continuum correction.

340 4.3. Calibration inversion

341 The aim is to retrieve the best parameters to simulate S_{sim} for each spectrum of each sequence of
 342 all solar fullscan. The goal is to compute the sensitivity (photometric calibration) but also to look
 343 at the potential correlation between the estimated free parameters and instrumental parameters
 344 such as temperature.

345 Since the AOTF parameters, the wavenumber shift, and the actual wavenumber resolution are
 346 not known and subject to change from one observation to another, we decided to estimate them
 347 by a fitting procedure. Except for the Gaussian and sinc-square functions intensities (the sum of
 348 which must be equal to 1, such that a filter is defined) all six other parameters are expressed in
 349 wavenumbers units (cm^{-1}) which makes it possible to assume the range in which we expect these
 350 parameters to fall. This is why we choose a bound-constrained minimization with the L-BFGS-B
 351 algorithm (Byrd et al., 1995; Zhu et al., 1997). The minimization equation follows:

$$\min_{I_0, fwhm, \Delta\nu, I_G, \sigma_G, \Delta_G, \delta\nu, \sigma_{ILS}} \left(\left\| S_{sim}^{smooth, resamp, flat}(I_0, fwhm, \Delta\nu, I_G, \sigma_G, \Delta_G, \delta\nu, \sigma_{ILS}) - S_{obs}^{flat, cal} \right\| \right) \quad (21)$$

352 The remaining residue between the spectra is estimated with the Root Mean Square Error
 353 (RMSE):

$$RMSE = \sqrt{\frac{\sum \left(S_{sim}^{smooth, resamp, flat} - S_{obs}^{flat, cal} \right)^2}{N}} \quad (22)$$

354 With N , the number of considered spectels, here 270 as the first 50 spectels are removed due
 355 to low SNR. The relative RMSE (rel_{RMSE}) can be defined as follow :

$$rel_{RMSE} = \frac{RMSE}{\overline{S_{sim}^{smooth, resamp, flat}}} \quad (23)$$

356 with $\overline{S_{sim}^{smooth, resamp, flat}}$, the mean of the simulation spectra. This gives the relative error (in %)
 357 between the observation and the simulation. The inversion stops when the norm of the projected
 358 gradient of the RMSE is less or equal to 10^{-8} with a step size of 10^{-5} for each parameter.

359 A total of 8 free parameters were retrieved for each spectrum of all sequences and all fullscans.

Parameter	Initialization	Range	Units
I_0	0.5	0.1 - 1.0	
fwhm	18	12.0 - 20.0	cm^{-1}
Δ_ν	0.1	-10.0 - 10.0	cm^{-1}
I_G	0.5	0.1 - 1.0	
σ_G	12	10.0 - 15.0	cm^{-1}
Δ_G	0.1	0.1 - 1.0	cm^{-1}
σ_ν	eq. 15	0.1 - 1.0	cm^{-1}
δ_ν	0.1	-2.0 - 2.0	cm^{-1}

Table 3: Summary of the initialization and boundaries used for the calibration inversion.

360 The minimization step needs an initialization of these parameters to look for the best fit. The
 361 initialization is crucial for a non-convex problem (which is the case here, since the problem is
 362 non-linear), that is why we tested several cases and finally reach a good optimum using the set of
 363 parameters for the initialization described in table 3.

364 5. Justification of the continuum removal

365 All modeled effects are not able to perfectly fit both the overall shape of the spectra and the
 366 absorption lines of the observation as shown in figure 5. In this figure, we applied the calibration
 367 procedure with/without the continuum removal (Fig. 5 A and B). One can see that without the
 368 continuum removal, the fitting procedure is dominated by the large-scale feature and unfortunately
 369 not coherent with detailed solar line shape. Such results imply that in the present state of instru-
 370 ment knowledge, it is not possible to take into account a precise instrumental transfer function for
 371 both shape and solar lines.

372 In order to check the consistency, we tried an alternative approach assuming perfectly known
 373 spectral calibration (Fig. 5 C). This may not be achievable in actual data analysis and would
 374 requires another pre-processing step of spectral calibration. Results show that the reconstruction
 375 is acceptable contrarily to the case without continuum removal (Fig. 5 B). Both continuum and
 376 solar lines can be reasonably well fitted, but the best reconstruction is for the fully blind case
 377 (inversion of the spectral calibration with continuum removal) (Fig. 5 A). As one of the main
 378 advantages of LNO being its very high spectral resolution, one would emphasize the spectral line
 379 reconstruction instead of the continuum shape.

380 Here three strategies are possible. The first approach is to improve the characterization of the

381 instrument, in particular by trying to handle the temperature dependence of the AOTF and grating
382 blaze function shape. Due to the small amount of in-flight solar data acquisition, this strategy
383 may be difficult, in particular to distinguish temperature dependence and temporal dependence
384 (due to the degradation of the detector in space for instance). In addition, an improvement of
385 the instrumental characterization using a spare ground instrument always rise the question of its
386 representativity.

387 A second approach would be purely empirical (on in-flight data only, no reference solar spec-
388 trum). By assuming negligible temporal variability, Thomas et al. (2021) propose a fully empirical
389 calibration, by estimating a comprehensive transfer function (including instrumental transfer func-
390 tion, sensitivity, and solar spectrum) and its temperature dependence, interpolated from actual
391 data. This strategy, based on observation only, may be biased due to the noise in the data (despite
392 an $\text{SNR} > 1000$) and possible temporal variability. Nevertheless, this approach has the advantage
393 to keep the large scale spectral features.

394 A third approach, proposed in the present article (see section 4.2.4), is to solve the problem by
395 removing the continuum, focusing on small-scale spectral features, but possibly erasing large-scale
396 spectral feature. This approach takes advantage, as previously mentioned, that the instrumental
397 transfer function is normalized, and thus filtering would not change the level. In addition, the
398 potential alteration of the large-scale feature would be identical for simulated and observed spectra,
399 and thus would produce no bias, as also performed on NOMAD-SO channel (Villanueva et al., 2021;
400 Knutsen et al., 2021).

401 The main advantage of such an approach is that it allows to keep the spectral radiance level
402 (which is of primary importance for absolute photometric calibration) and produce flat continuum
403 spectra. The validity of this approach can be measured by the root-mean-square error (RMSE)
404 that should be low in comparison to the signal, but also by the coherence of the sensitivity mea-
405 surement (consecutive measurement should be somehow similar). We propose hereafter to follow
406 this strategy, which has the advantage to avoid any assumption of the temporal stability of the
407 NOMAD instrument.

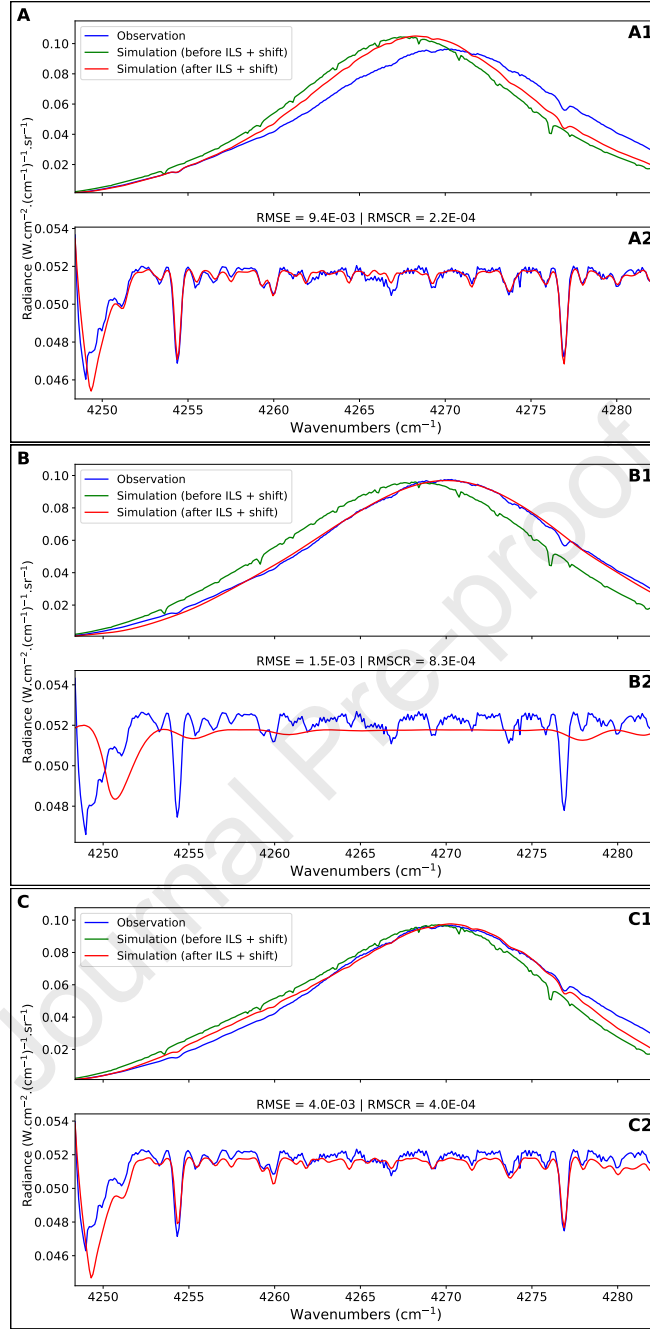


Figure 5: (A) Estimation of the inversion with continuum removal using the minimization equation 21. Note that the first 50 spectels on the left (up to 4253.5 cm^{-1}) are not included in the inversion due to low signal to noise ratio. (A1) Comparison between an observation ($S_{obs}(\nu)$, in blue) and two types of inversion: before integration of the wavenumber shift and ILS ($S_{sim}^{instru}(\nu)$, in green) and after ($S_{sim}^{smooth, resamp}(\nu)$, in red). The RMSE indicates the quality of the fit. (A2) Comparison between an observation (S_{obs}^{flat} , in blue) and the simulation ($S_{sim}^{smooth, resamp, flat}(\nu)$, in red) both after continuum removal. The RMSCR (RMSE after continuum removal) indicates the quality of the fit. (B) Same as (A) without continuum removal. (C) same as (A) without continuum removal but with spectral calibration assumed to be perfectly known. For the (B2) and (C2) plots we compute the continuum removal on the results of the inversion (continuum removal was not included in the inversion).

408 6. Results

409 *Fit of order 189.* An interesting way to check the validity of the model is to compare observations
 410 and simulations at the spectrum scale. Figure 6 shows such comparison for order 189 on the whole
 411 of a fullscan (here 14/03/2019), as an example. Here the spectra are flat due to the continuum
 412 removal step which allows apprehending only the level of the spectrum and the position of the
 413 bands. The 13 observation sequences are shown in a shade of gray while the optimized simulation
 414 after the inversion is shown in colors. The residue between the observation and the simulation is
 415 also indicated. Both the main and weak solar lines are well reproduced and the small discrepancies
 416 are within the continuum and related to very weak solar lines. As the position and depth of the
 417 line are mainly controlled by the AOTF transfer function, with the secondary lobes of the transfer
 418 function being responsible for the leakage of adjacent orders on the central order (here 189).
 419 This contribution is expected to significantly increase the intensity of such solar lines, especially
 420 if they are located on the edges of the order, as they can be duplicated. Order 189 displays
 421 such potentiality with two main solar lines on both edges of the detector. One can see that the
 422 model perfectly accounts for this contribution since the simulation match very well the observation
 423 (relative RMS = 0.43%), no matter the observation sequence.

424 *RMSE.* Here we illustrate the overall results of the calibration model. First, the RMSE can be
 425 used for all observations as it is the quality criterion of the approach. Figure 7 shows the relative
 426 RMSE based on equation 23 for diffraction orders 110 to 200. The large majority of the points are
 427 below 1% error except for extreme orders (100-110 and 198-200) for which the SNR can be very
 428 low. We can also notice that the fullscan showing the widest dispersion is that of 01/11/2018 for
 429 which the temperature during the observation was abnormally high, leading to noisier spectra and
 430 therefore greater difficulty in obtaining a correct fit. The relative RMSE appears more dispersed
 431 for orders 150 to 190 compared to orders 120 to 150. This can be explained by the fact that
 432 high orders have stronger solar lines, which implies more information in the spectra to model
 433 the AOTF resulting in a larger standard deviation of the RMSE. However, if we exclude outliers
 434 above 1% , the average value of the relative RMSE between orders 150-190 is comparable to that
 435 of orders 120-150. This low RMSE for most of the orders (around 0.5%) tends to validate the

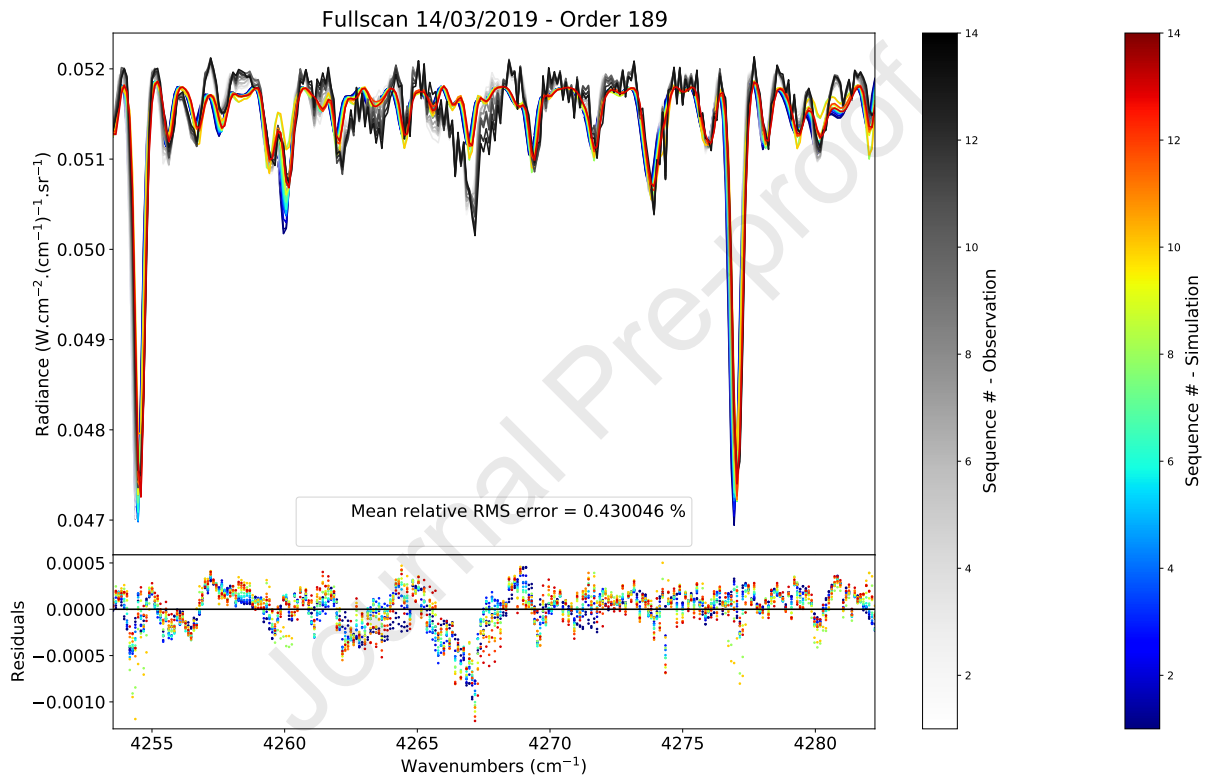


Figure 6: Fullscan of 14/03/2019, comparison between all observation sequences of order 189 (in shades of gray) and the respective optimized simulation (in colors). One can see that the main solar lines are well reproduced and the small discrepancies are only located within the continuum, near the weak solar lines. The mean relative RMS error of the 13 sequences is 0.43%.

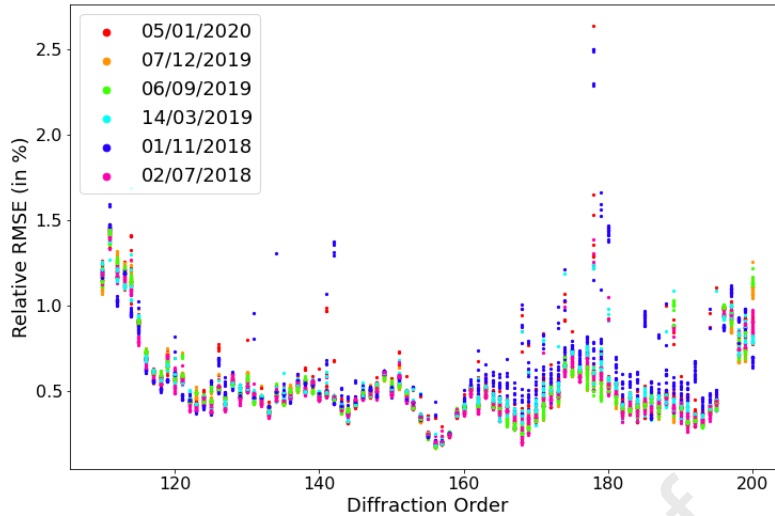


Figure 7: Relative RMS error rel_{RMSE} between the simulation and the observation for diffraction orders 110 to 200 and the 6 solar fullscans. Each point represents a single spectrum of a fullscan sequence.

436 spectrum-to-spectrum calibration method using a continuum removal approach.

437 *Sensitivity.* The main objective of the model is to establish the conversion factor from the measured
 438 ADU by the detector to spectral radiance. This step is done using equation 19 on all spectra of all
 439 fullscan so that we can look at the potential variations of this factor during the mission. Figure
 440 8 shows the sensitivity curves of the LNO channel, which was estimated using equation 19. As
 441 we estimate the mean *Sensitivity* on 270 spectels, it expresses the mean radiance per counts per
 442 second of this 270 spectels. The shape of the curves is very similar to what was obtained during
 443 ground calibration before the mission: orders 120 to 135 and 165 to 190 have a higher sensitivities
 444 (Thomas, 2020). This is related to the initial design of the instrument whose purpose was a precise
 445 atmospheric characterization and these orders match the spectral range in which CO, H₂O and CH₄
 446 have significant absorption. In addition, a drastic fall of the sensitivity near order 150 was expected
 447 and is correctly reproduced. The significant drop of the sensitivity for extreme orders illustrates
 448 well how the SNR can be low for these orders, and therefore the reasons for a higher RMSE (noisy
 449 observation are harder to fit). An exception is made for orders around 150 as the sensitivity is
 450 quite low but the RMSE of these orders is also the lowest. This is related to the fact that these
 451 orders have very few solar lines which imply an easier fit of an almost flat continuum. Nevertheless,
 452 from one fullscan to another the shape is constant but the intensity slightly changes. The first
 453 two fullscans (02/07/2018 in purple and 01/11/2011 in navy blue) display lower sensitivities while

454 fullscans of 06/09/2019 (in green) and 07/12/2019 (in orange) show the highest values. These
 455 differences can be explained in many ways and one of the advantages of this calibration method
 456 is to be able to test the dependence of the sensitivity such as a function of time, of the Mars-Sun
 457 distance, or the temperature. Hereafter we will demonstrate that the temperature dependence is
 458 the dominant factor.

459 With such variation of the instrumental sensitivity (Fig. 8), one would look if there is any
 460 correlation with other parameters. This step was made using both the nominal and redundant
 461 temperature T sensors of the LNO that monitors the temperature within the spectrometer every
 462 2 minutes. Here we show the correlation with the nominal sensor. The temporal resolution of the
 463 temperature sensor is 2 minutes, which is too high compared to the time between two consecutive
 464 LNO observations. Temperature data were thus interpolated to match the timestamp of each
 465 fullscan. Figure 9 shows the sensitivity as a function of the temperature. Here all diffraction orders
 466 of the six fullscans are shown along with the linear regression (dotted line). The gap between the
 467 points illustrates the temperature variation between the fullscans. Within one sequence, one can
 468 see a slope that seems to be consistent between all orders showing that sensitivity decline when
 469 temperature increases. Looking at all sequences together, the relationship seems linear and we fit
 470 a line for each diffraction order:

$$\text{Sensitivity} = a.T + b \quad (24)$$

471 With T the temperature. The adjusted a and b values are available in public archive¹. By
 472 postulating that the temperature is mainly responsible for the sensitivity change of the detector,
 473 and correcting these variations with a linear fit we can show that there is no sign of a correlation
 474 with time (aging of the detector) in the instrumental response of the detector (see figure 10) which
 475 was not obvious in the first place.

476 Regarding the absolute uncertainty of the sensitivity retrieved here, and hence on the radiance
 477 derived from the calibration. All the steps introduced in the calibration pipeline (ILS, spectral
 478 shift, AOTF parameters and continuum correction) respect the energy conservation. Thus, ac-

¹<https://doi.org/10.14768/2d7688ba-2e5f-473c-8e03-763e93f4bd6b>

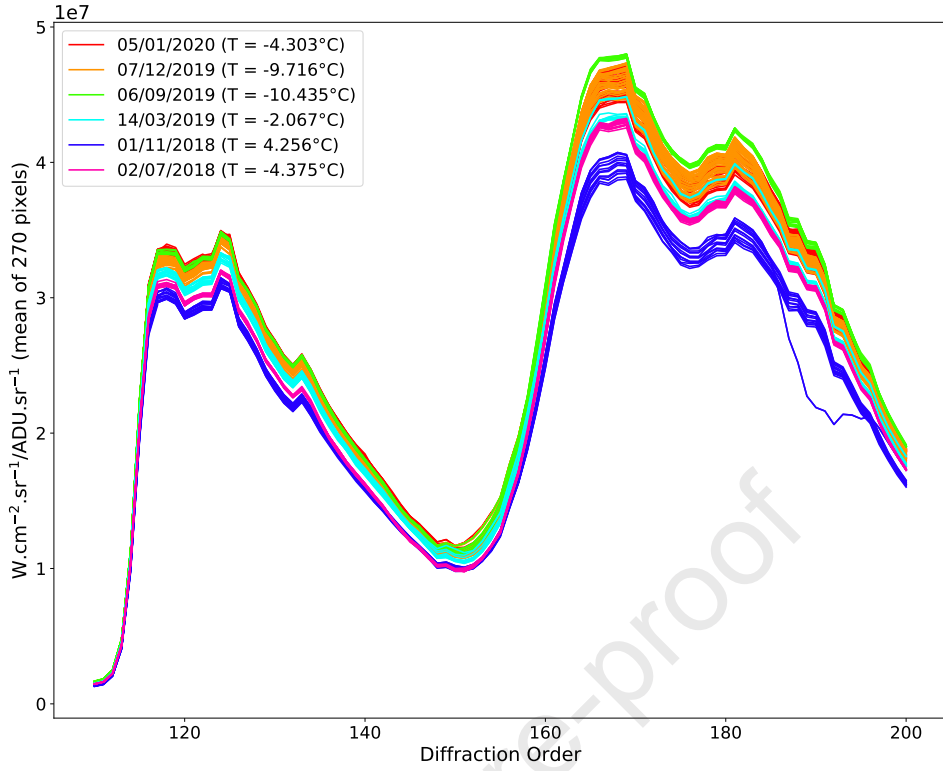


Figure 8: Radiance to counts conversion (*Sensitivity*) for diffraction orders 110 to 200 retrieved for the 6 LNO solar fullscan using equation 19. The mean temperature during the fullscan is indicated. This illustrates the sensitivity curves of the LNO channel. The shape is very consistent from one sequence to another but also between the fullscans, with two main peaks of sensitivity around orders 125 and 170 and a lower sensitivity for the extreme orders and around orders 150.

479 cording to equation 19, the main source of uncertainty come from the number of counts (S_{obs}^{flat}
 480 in $ADU.(cm^{-1})^{-1}.s^{-1}$) of the LNO observation of the sun and from the synthetic solar spectrum
 481 ($S_{sim}^{smooth,resamp,flat}$ in spectral radiance). The LNO observation are normalized according to equa-
 482 tion 1 and we can assume that the uncertainty on the integration time, number of accumulation,
 483 spectral resolution and binning factor is negligible. According to Thomas et al. (2021), a typical
 484 LNO spectrum of 3000 ± 500 counts would have an SNR of about 25 ± 5 , which gives an absolute
 485 error between 0.6% and 1%. Also, the maximum expected systematic error is 4% (Thomas et al.,
 486 2021). The uncertainty on $S_{sim}^{smooth,resamp,flat}$ is related to the initial solar spectrum, according to
 487 Chance and Kurucz (2010), the absolute accuracy of the solar spectrum is of 3.5-4%. This leads
 488 to a maximum absolute uncertainty on the radiance less than 10%. Although the maximum error
 489 may be high, the systematic error of LNO is not often expected to be 4%.

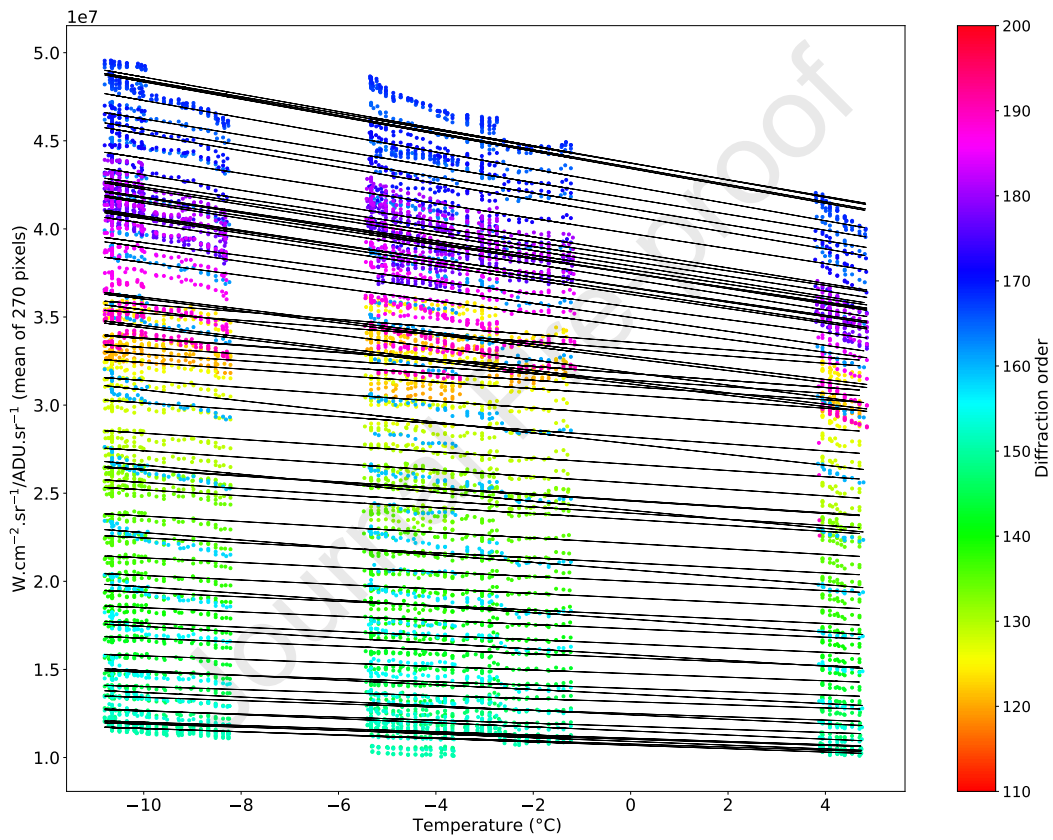


Figure 9: Sensitivity as a function of the instrumental temperature (nominal sensor). The colors refer to diffraction order and the dotted line shows the linear regression per order.

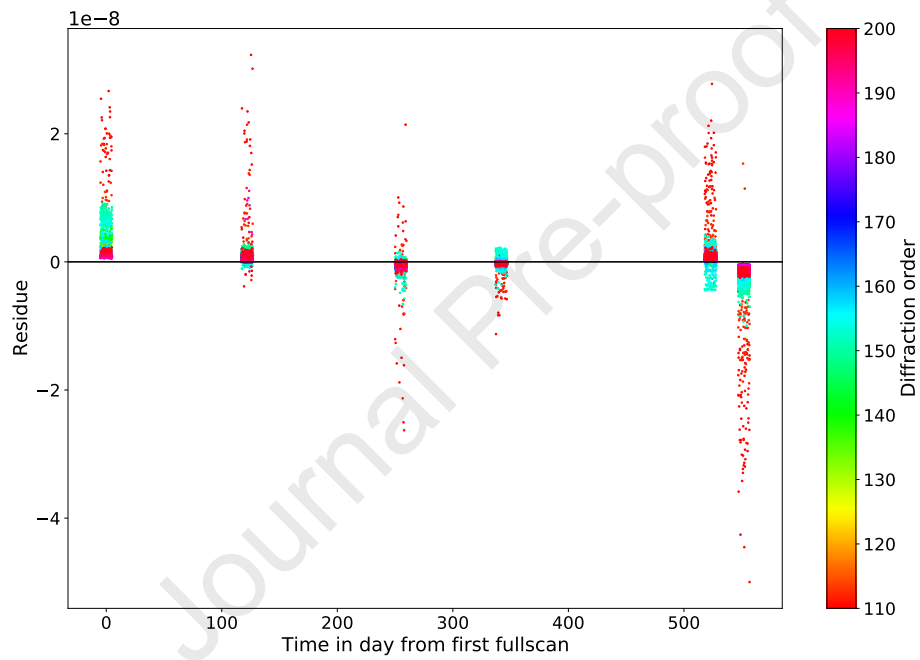


Figure 10: Residue after temperature correction of the sensitivity using eq. 24 as a function of time (in day) from first fullscan. The results are artificially spread horizontally for clarity. Despite few outliers near order 110 (in red) for which the sensitivity is very low, the residues indicate that there is no correlation of the sensitivity with time and therefore no aging of the detector to the first order.

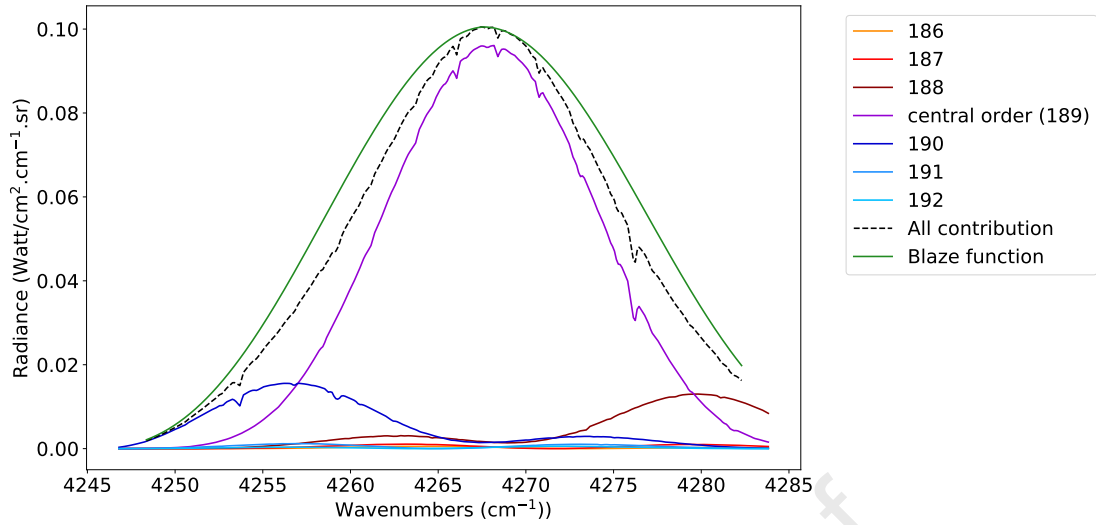


Figure 11: AOTF transfer function: contribution from the main order (189 in purple) and 3 adjacent orders. The sum of all contributions that define the final spectrum is shown in black dashed line and the Blaze function of the central order is shown in green. The main lobe of the AOTF transfer function accounts for the contribution of the central order while the sidelobes control the shape and intensity of adjacent orders which are summed onto the wavenumbers of the central order. The shape of the final spectrum is therefore constrained by the product of the Blaze function and the AOTF transfer function.

490 *AOTF*. Among all the inverted parameters six of them account for the AOTF transfer function
 491 shape. They are crucial because they account for the contribution of the main and adjacent orders
 492 which define the intensity of solar lines. An example of the transfer function retrieved after the
 493 inversion is given in figure 11 for one observation of order 189. In this case, the estimated AOTF
 494 transfer function is slightly asymmetrical, promoting a higher contribution of order 190 compared
 495 to order 188. The “bell” shape of an LNO spectrum is partly controlled by the blaze function
 496 but the AOTF sidelobes are also affecting the spectra by widening or shrinking the flanks of the
 497 continuum. To reproduce all the potential shapes of an LNO spectrum, which vary from one
 498 order to another it is necessary to integrate variations in intensity, width and offsets of the AOTF
 499 transfer function.

500 In the example shown figure 11 the AOTF inverted parameters to reconstruct the final spectrum
 501 are:

- 502 • $I_0 = 0.74 \pm 0.3$
- 503 • $fwhm = 17.41 \text{ cm}^{-1} \pm 0.86$
- 504 • $\Delta_\nu = 2.34 \text{ cm}^{-1} \pm 0.53$

- 505 • $I_G = 0.71 \pm 0.57$
- 506 • $\sigma_G = 12.86 \text{ cm}^{-1} \pm 0.77$
- 507 • $\Delta_G = 2.33 \text{ cm}^{-1} \pm 0.59$

508 We tested several temperature and temporal dependence and did not find any obvious correlation.
 509 These parameters define the shape of the AOTF and can therefore be strongly correlated together.
 510 This is particularly noticeable on the spectral shift of the sinc-square (Δ_ν) and the gaussian (Δ_G)
 511 whose values and uncertainties are similar. A shift of one can thus be compensated by the other,
 512 but the best fit is estimated for a similar offset of the two. This observation is the same for the
 513 intensities of the two filters (I_0 and I_G) whose uncertainties are relatively large. However, the
 514 width of the filters ($fwhm$ and σ_G) is well constrained and the values obtained are similar to what
 515 is obtained by Liuzzi et al. (2019).

516 *Spectral calibration and ILS.* Figure 12 shows the actual spectral resolution of the instrument σ_ν
 517 retrieved by the algorithm for all available data as a function of the diffraction order. A comparison
 518 with the theoretical spectral resolution computed using eq. 15 is provided. We found that σ_ν is
 519 slightly higher (mostly around 0.05) than expected but still follows the theoretical trend. Some
 520 points are way off the trend and are often related to orders with lower sensitivity and thus more
 521 noise, so the convolution requires a much wider ILS to get a proper fit. This is illustrated with
 522 the fullscan of 01/11/2018 (navy blue points), which corresponds to the highest temperature and
 523 lowest sensitivity (see fig. 9, navy blue curves)

524 The wavenumber shift δ_ν account for potential deviation from the theoretical spectral calibra-
 525 tion (see eq. 4). Figure 13 (top panel) shows the retrieved shift as a function of the diffraction
 526 order for the 6 fullscans. When expressed in pixel, the shift appears constant with the diffraction
 527 order. Nevertheless, the shifting intensity is different from one fullscan to another. We tested a
 528 temperature dependence of the wavenumber shift δ_ν in figure 13 (bottom panel) using orders 130
 529 to 190 and found a slope of about 0.75 pixels per degree which is consistent with what Liuzzi et al.
 530 (2019) and Thomas et al. (2021) reported with values of 0.71 and 0.83 pixels per degree.

531 The wavenumber shift and ILS analysis emphasize that our approach works well for diffraction

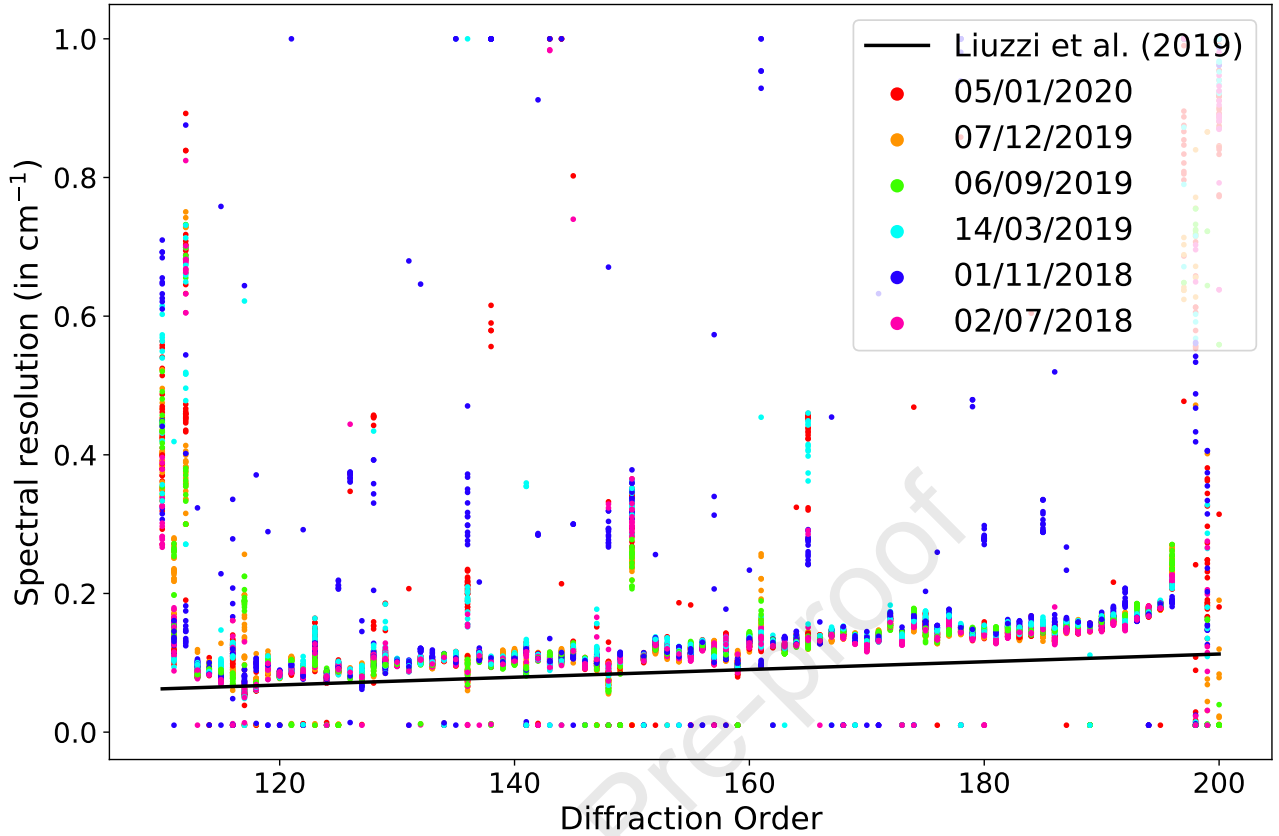


Figure 12: Actual ILS spectral resolution σ_ν retrieved from the calibration inversion. Each point corresponds to a single spectrum, the color refers to the 6 fullscans used. The empirical spectral resolution computed using eq. 15 (Liuzzi et al., 2019)) is shown as a black line. For a clearer comparison, data from Liuzzi et al. (2019) were converted from a Gaussian FWHM to a Gaussian standard deviation considering $FWHM = 2\sqrt{2 \ln 2} \sigma$

532 orders with multiple well-defined solar lines because the continuum removal approach allows as-
 533 signing a greater weight to the variations brought by the solar lines and therefore to the AOTF
 534 parameters. The quality of the fit is higher for orders with the number of strong solar lines. Also,
 535 such orders have often higher sensitivities which promote a better estimation of the wavenumber
 536 shift and the spectral resolution.

537 7. Calibration pipeline

538 With such an approach any nadir observation can be calibrated to spectral radiance, the pipeline
 539 to calibrate a raw nadir spectrum is as follows: first, the raw spectrum is normalized following
 540 equation 1 with the spectral resolution (eq. 15). Second, we remove the continuum using eq. 18
 541 to get a flat spectrum. Then, knowing the temperature at the time of the measurement from
 542 housekeeping and using coefficients a and b from eq. 24, we apply the sensitivity factor to convert

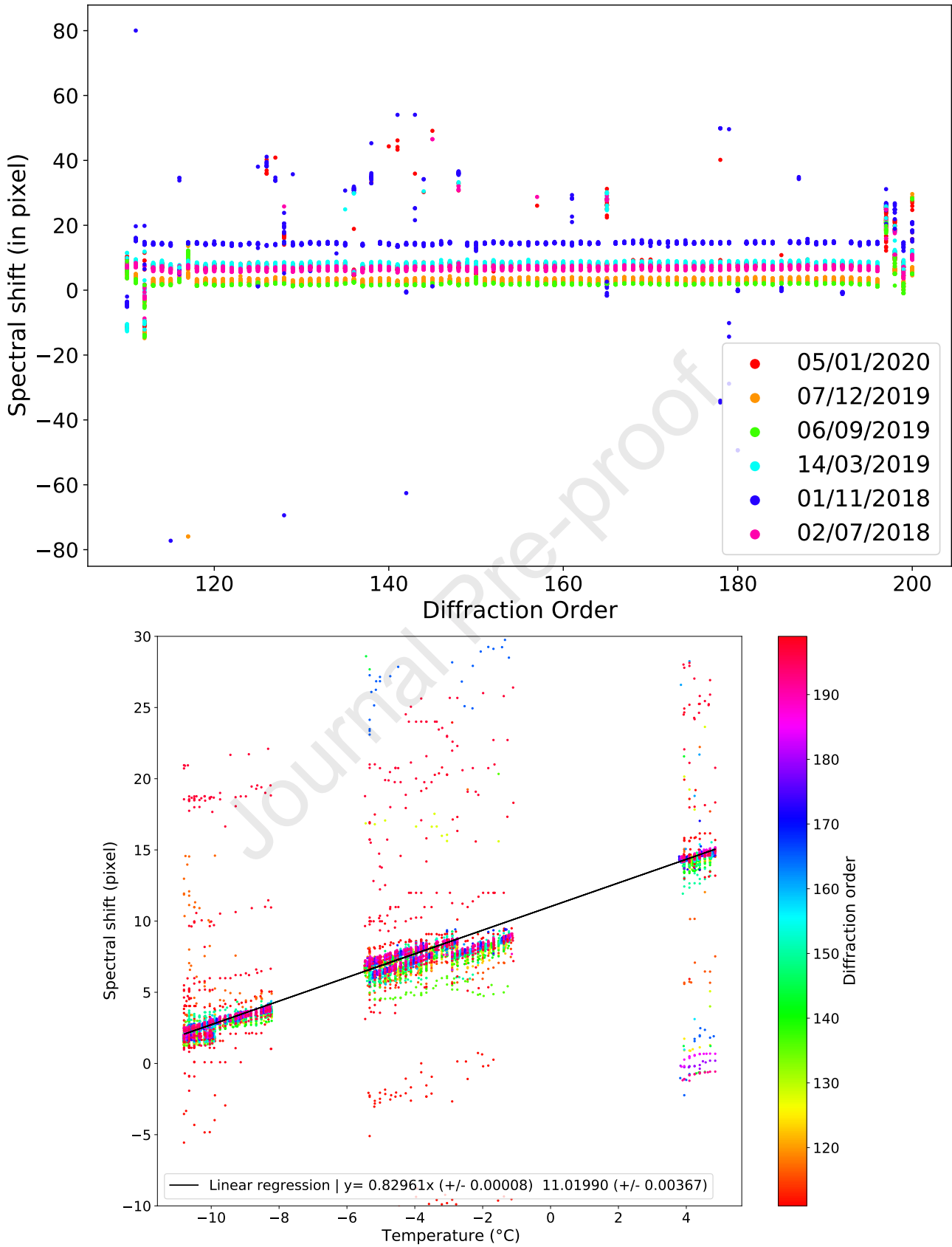


Figure 13: Wavenumber shift δ_ν (in pixel units) retrieved from the calibration inversion. Top image show the spectral shift as a function of the diffraction order, colors refer to the fullscan. Bottom image shows the shift as a function of temperature. The color refers to the diffraction order.

543 the normalized ADU to spectral radiance (see Eq. 20). This product can be used to create
 544 reflectance map such as the one reported in Thomas et al. (2021) applying the following equation:

$$R = \pi \cdot \frac{I}{F} \quad (25)$$

545 With I the spectral radiance and $F = S_L \cdot \pi \cdot \cos(\theta_i)$ the spectral irradiance received by the
 546 surface (see section 3.2) accounting for the incident angle θ_i and the distance of the sun at the
 547 time of the observation. A typical reflectance map obtained via our approach is shown in figure 14
 548 for order 189 using data spread over mostly 3 years. A comparison with the reflectance obtained
 549 following the method of Thomas et al. (2021) is also shown (fig. 14, bottom panel). The two
 550 calibrations provide very similar results with a linear coefficient of 1.024 and -0.017 (correlation
 551 coefficient of 0.987) despite different approaches. This result shows that both calibrations are in
 552 agreement mostly within 3% and that our approach underestimates (resp. overestimates) the one
 553 from Thomas et al., 2011 for reflectance < 0.2 (resp. > 0.2).

554 8. Conclusion

555 We propose an alternative calibration method for the LNO data using reference solar spectra
 556 with the advantage of being able to investigate the correlations between the instrumental sensitivity
 557 and the temperature of the instrument. By having done this, we can understand the potential
 558 temporal variations of the instrument due to its aging. The method is based on the adjustment of a
 559 synthetic spectrum to the solar data acquired with NOMAD-LNO fullscan operation mode, which
 560 allows a calibration over the whole spectral range of the instrument (different orders), using an
 561 analytical model that simulate the instrumental effects. To realize this calibration, we propose to
 562 focus on the level and the fine solar bands shape and remove the continuum. This strategy allows
 563 to correct of unknowns in the instrumental AOTF/blaze functions but could possibly remove the
 564 large spectral feature of the order. However, by performing the same continuum removal to the
 565 synthetic data, it is possible to compare synthetic and observation data without bias (Villanueva
 566 et al., 2021; Knutsen et al., 2021).

567 In addition, this method makes it possible to test the correlations between sensitivity and
 568 instrumental parameters (AOTF shape, ILS, wavenumber shift), such as temperature and time.

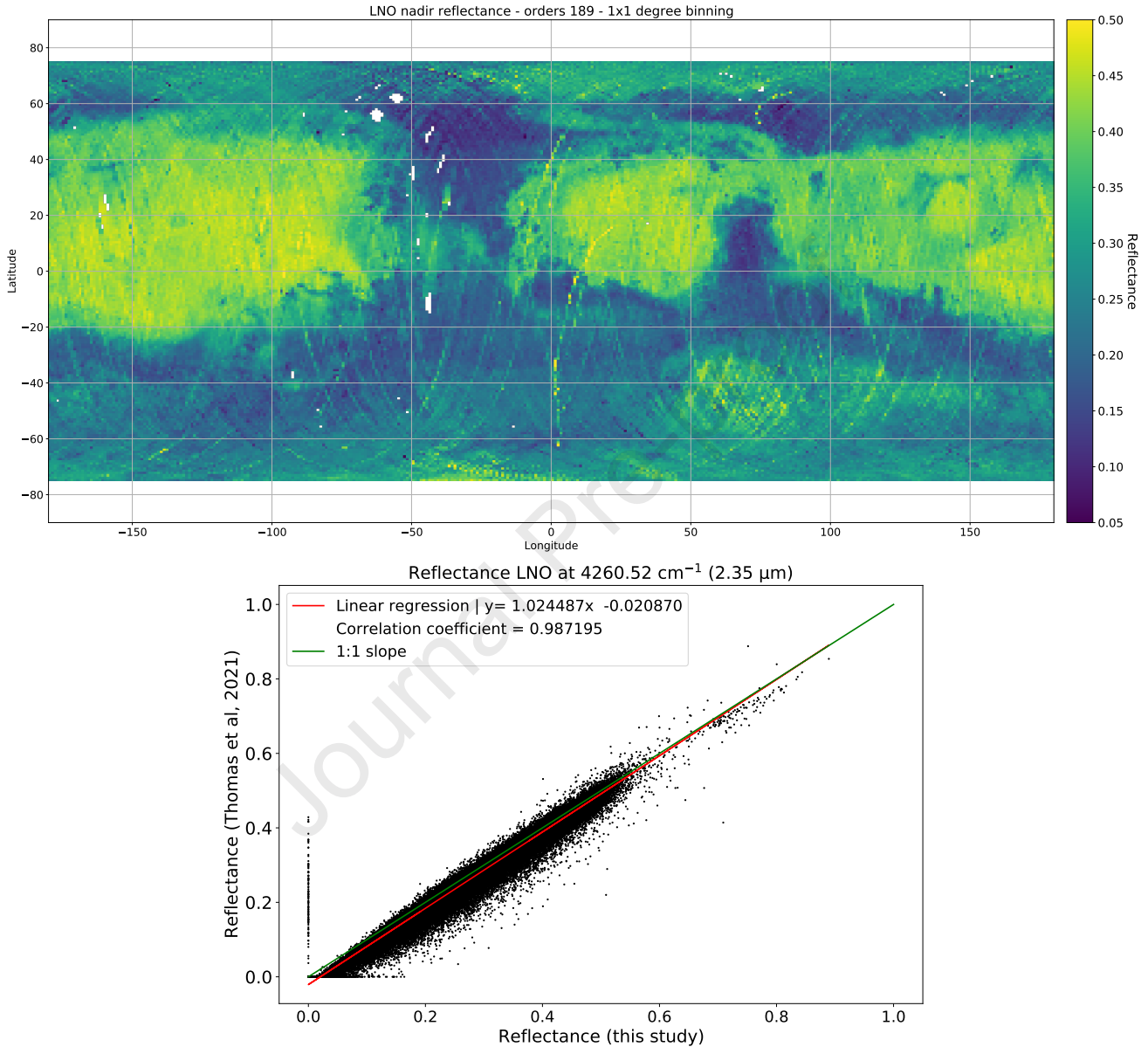


Figure 14: Reflectance of order 189 at 4260.52 cm^{-1} using 1230 orbits from 29/03/2018 to 19/10/2021. Data with incident angle higher than 80° were removed to promote a better SNR. Also, to reduce albedo variation due to seasonal effects at high latitude we only considered data between latitude 75° N and 75° S . The top image shows a map binned onto a $1^\circ \times 1^\circ$ grid using 134,738 spectra. Bottom shows a comparison between the reflectance estimate in this study and the reflectance obtained using the approach of Thomas et al. (2021) for the same order.

569 By doing this, we show that there is at first degree a linear relationship between the temperature
570 and the sensitivity. By removing this effect, we demonstrated that there is no sign of temporal
571 change in the sensitivity, which validates the calibration method proposed by Thomas et al. (2021).
572 If any temporal variations of the detector occur in the future, our method should be able to capture
573 them.

574 Finally, with such a model, one can calibrate any nadir observation to spectral radiance and
575 reflectance data. The calibrated data show a very strong correlation with the approach proposed
576 by Thomas et al. (2021) mostly within 3%. The two approaches are therefore complementary and
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- 596 Byrd, R.H., Lu, P., Nocedal, J., Zhu, C., 1995. A limited memory algorithm for bound constrained
597 optimization. *SIAM Journal on Scientific Computing* 16, 1190–1208. doi:10.1137/0916069.
- 598 Chance, K., Kurucz, R., 2010. An improved high-resolution solar reference spectrum for earth's at-
599 mosphere measurements in the ultraviolet and visible and and near infrared. *Journal of Quantita-
600 tive Spectroscopy and Radiative Transfer* 111, 1289–1295. doi:10.1016/j.jqsrt.2010.01.036.
- 601 Eilers, Paul, Boelens, Hans, 2005. Baseline corection with asymmetric least squares smoothing.
- 602 Hase, F., Wallace, L., McLeod, S.D., Harrison, J.J., Bernath, P.F., 2010. The ACE-FTS atlas of
603 the infrared solar spectrum. *Journal of Quantitative Spectroscopy and Radiative Transfer* 111,
604 521–528. doi:10.1016/j.jqsrt.2009.10.020.
- 605 Knutsen, E.W., Villanueva, G.L., Liuzzi, G., Crismani, M.M., Mumma, M.J., Smith, M.D., Van-
606 daele, A.C., Aoki, S., Thomas, I.R., Daerden, F., Viscardy, S., Erwin, J.T., Trompet, L., Neary,
607 L., Ristic, B., Lopez-Valverde, M.A., Lopez-Moreno, J.J., Patel, M.R., Karatekin, O., Bellucci,
608 G., 2021. Comprehensive investigation of mars methane and organics with ExoMars/NOMAD.
609 *Icarus* 357, 114266. doi:10.1016/j.icarus.2020.114266.
- 610 Liuzzi, G., Villanueva, G.L., Mumma, M.J., Smith, M.D., Daerden, F., Ristic, B., Thomas, I.,
611 Vandaele, A.C., Patel, M.R., Lopez-Moreno, J.J., Bellucci, G., 2019. Methane on mars: New
612 insights into the sensitivity of CH₄ with the NOMAD/ExoMars spectrometer through its first
613 in-flight calibration. *Icarus* 321, 671–690. doi:10.1016/j.icarus.2018.09.021.
- 614 Marquardt, E., Le, J., Radebaugh, R., 2000. Cryogenic material properties database. 11th Inter-
615 national Cryocooler Conference .
- 616 Neefs, E., Vandaele, A.C., Drummond, R., Thomas, I.R., Berkenbosch, S., Clairquin, R., Delanoye,
617 S., Ristic, B., Maes, J., Bonnewijn, S., Pieck, G., Equeter, E., Depiesse, C., Daerden, F.,
618 Ransbeeck, E.V., Nevejans, D., Rodriguez-Gómez, J., López-Moreno, J.J., Sanz, R., Morales,
619 R., Candini, G.P., Pastor-Morales, M.C., del Moral, B.A., Jeronimo-Zafra, J.M., Gómez-López,
620 J.M., Alonso-Rodrigo, G., Pérez-Grande, I., Cubas, J., Gomez-Sanjuan, A.M., Navarro-Medina,
621 F., Thibert, T., Patel, M.R., Bellucci, G., Vos, L.D., Lesschaeve, S., Vooren, N.V., Moelans, W.,

- 622 Aballea, L., Glorieux, S., Baeke, A., Kendall, D., Neef, J.D., Soenen, A., Puech, P.Y., Ward,
623 J., Jamoye, J.F., Diez, D., Vicario-Arroyo, A., Jankowski, M., 2015. NOMAD spectrometer on
624 the ExoMars trace gas orbiter mission: part 1—design and manufacturing and testing of the
625 infrared channels. *Applied Optics* 54, 8494. doi:10.1364/ao.54.008494.
- 626 Nevejans, D., Neefs, E., Ransbeeck, E.V., Berkenbosch, S., Clairquin, R., Vos, L.D., Moelans,
627 W., Glorieux, S., Baeke, A., Korablev, O., Vinogradov, I., Kalinnikov, Y., Bach, B., Dubois,
628 J.P., Villard, E., 2006. Compact high-resolution spaceborne echelle grating spectrometer with
629 acousto-optical tunable filter based order sorting for the infrared domain from 2.2 to 4.3 μm .
630 *Applied Optics* 45, 5191. doi:10.1364/ao.45.005191.
- 631 Smith, M., 2021. Nomad nadir-geometry retrievals of co in the mars atmosphere. doi:10.17632/
632 PX89DK6CK9.1.
- 633 Thomas, I., 2020. Nomad experiment to archive interface control document. URL: ftp:
634 //npsa01.esac.esa.int/pub/mirror/ExoMars2016/em16_tgo_nmd/document/EAICDhttps:
635 //nomad.aeronomie.be/ProjectDir/data/nomad_hdf5_datasets.pdf.
- 636 Thomas, I., Aoki, S., Trompet, L., Robert, S., Depiesse, C., Willame, Y., Cruz-Mermy, G.,
637 Schmidt, F., Erwin, J., Vandaele, A., Daerden, F., Mahieux, A., Neefs, E., Ristic, B., Hetey, L.,
638 Patel, M., Moreno, J.L., Bellucci, G., , the NOMAD Team, 2021. Calibration of nomad onboard
639 the exomars trace gas orbiter: Part 2 - the lno channel. Submitted .
- 640 Thomas, I.R., Vandaele, A., Robert, S., Neefs, E., Drummond, R., Daerden, F., Delanoye, S.,
641 Ristic, B., Berkenbosch, S., Clairquin, R., Maes, J., Bonnewijn, S., Depiesse, C., Mahieux, A.,
642 Trompet, L., Neary, L., Willame, Y., Wilquet, V., Nevejans, D., Aballea, L., Moelans, W.,
643 Vos, L.D., Lesschaeve, S., Vooren, N.V., Lopez-Moreno, J.J., Patel, M.R., Bellucci, G., the
644 NOMAD Team, 2016. Optical and radiometric models of the NOMAD instrument part II: the
645 infrared channels - SO and LNO. *Optics Express* 24, 3790. doi:10.1364/oe.24.003790.
- 646 Vago, J., Witasse, O., Svedhem, H., Baglioni, P., Haldemann, A., Gianfiglio, G., Blancquaert, T.,

647 McCoy, D., de Groot, R., 2015. ESA ExoMars program: The next step in exploring mars. *Solar*
648 *System Research* 49, 518–528. doi:10.1134/s0038094615070199.

649 Vandaele, A., Neefs, E., Drummond, R., Thomas, I., Daerden, F., Lopez-Moreno, J.J., Rodriguez,
650 J., Patel, M., Bellucci, G., Allen, M., Altieri, F., Bolsée, D., Clancy, T., Delanoye, S., Depiesse,
651 C., Cloutis, E., Fedorova, A., Formisano, V., Funke, B., Fussen, D., Geminale, A., Gérard, J.C.,
652 Giuranna, M., Ignatiev, N., Kaminski, J., Karatekin, O., Lefèvre, F., López-Puertas, M., López-
653 Valverde, M., Mahieux, A., McConnell, J., Mumma, M., Neary, L., Renotte, E., Ristic, B.,
654 Robert, S., Smith, M., Trokhimovsky, S., Auwera, J.V., Villanueva, G., Whiteway, J., Wilquet,
655 V., Wolff, M., 2015. Science objectives and performances of NOMAD and a spectrometer suite
656 for the ExoMars TGO mission. *Planetary and Space Science* 119, 233–249. doi:10.1016/j.pss.
657 2015.10.003.

658 Vandaele, A.C., , Korablev, O., Daerden, F., Aoki, S., Thomas, I.R., Altieri, F., López-Valverde,
659 M., Villanueva, G., Liuzzi, G., Smith, M.D., Erwin, J.T., Trompet, L., Fedorova, A.A.,
660 Montmessin, F., Trokhimovskiy, A., Belyaev, D.A., Ignatiev, N.I., Luginin, M., Olsen, K.S.,
661 Baggio, L., Alday, J., Bertaux, J.L., Betsis, D., Bolsée, D., Clancy, R.T., Cloutis, E., Depiesse,
662 C., Funke, B., Garcia-Comas, M., Gérard, J.C., Giuranna, M., Gonzalez-Galindo, F., Grigoriev,
663 A.V., Ivanov, Y.S., Kaminski, J., Karatekin, O., Lefèvre, F., Lewis, S., López-Puertas, M.,
664 Mahieux, A., Maslov, I., Mason, J., Mumma, M.J., Neary, L., Neefs, E., Patrakeevev, A., Patsaev,
665 D., Ristic, B., Robert, S., Schmidt, F., Shakun, A., Teanby, N.A., Viscardy, S., Willame, Y.,
666 Whiteway, J., Wilquet, V., Wolff, M.J., Bellucci, G., Patel, M.R., López-Moreno, J.J., For-
667 get, F., Wilson, C.F., Svedhem, H., Vago, J.L., and, D.R., 2019. Martian dust storm impact
668 on atmospheric h₂o and d/h observed by ExoMars trace gas orbiter. *Nature* 568, 521–525.
669 doi:10.1038/s41586-019-1097-3.

670 Vandaele, A.C., , Lopez-Moreno, J.J., Patel, M.R., Bellucci, G., Daerden, F., Ristic, B., Robert,
671 S., Thomas, I.R., Wilquet, V., Allen, M., Alonso-Rodrigo, G., Altieri, F., Aoki, S., Bolsée,
672 D., Clancy, T., Cloutis, E., Depiesse, C., Drummond, R., Fedorova, A., Formisano, V., Funke,
673 B., González-Galindo, F., Geminale, A., Gérard, J.C., Giuranna, M., Hetey, L., Ignatiev, N.,

674 Kaminski, J., Karatekin, O., Kasaba, Y., Leese, M., Lefèvre, F., Lewis, S.R., López-Puertas,
675 M., López-Valverde, M., Mahieux, A., Mason, J., McConnell, J., Mumma, M., Neary, L., Neefs,
676 E., Renotte, E., Rodriguez-Gomez, J., Sindoni, G., Smith, M., Stiepen, A., Trokhimovsky,
677 A., Auwera, J.V., Villanueva, G., Viscardy, S., Whiteway, J., Willame, Y., Wolff, M., 2018.
678 NOMAD and an integrated suite of three spectrometers for the ExoMars trace gas mission:
679 Technical description and science objectives and expected performance. *Space Science Reviews*
680 214. doi:10.1007/s11214-018-0517-2.

681 Vandaele, A.C., Mahieux, A., Robert, S., Berkenbosch, S., Clairquin, R., Drummond, R., Letocart,
682 V., Neefs, E., Ristic, B., Wilquet, V., Colomer, F., Belyaev, D., Bertaux, J.L., 2013. Improved
683 calibration of SOIR/venus express spectra. *Optics Express* 21, 21148. doi:10.1364/oe.21.
684 021148.

685 Villanueva, G.L., Liuzzi, G., Crismani, M.M.J., Aoki, S., Vandaele, A.C., Daerden, F., Smith,
686 M.D., Mumma, M.J., Knutsen, E.W., Neary, L., Viscardy, S., Thomas, I.R., Lopez-Valverde,
687 M.A., Ristic, B., Patel, M.R., Holmes, J.A., Bellucci, G., and, J.J.L.M., 2021. Water heavily
688 fractionated as it ascends on mars as revealed by ExoMars/NOMAD. *Science Advances* 7,
689 eabc8843. doi:10.1126/sciadv.abc8843.

690 Zhu, C., Byrd, R.H., Lu, P., Nocedal, J., 1997. Algorithm 778: L-BFGS-b: Fortran subroutines
691 for large-scale bound-constrained optimization. *ACM Transactions on Mathematical Software*
692 23, 550–560. doi:10.1145/279232.279236.

Highlights

The LNO channel is 1 of 3 spectrometers in NOMAD on ExoMars Trace Gas Orbiter.

The LNO channel observes Mars in limb, nadir and solar occultation modes

This article describes an alternative method of calibration for the LNO channel data based on a comparison between observation of the sun by the LNO channel and a synthetic solar spectrum

Temperature and time dependence of the instrumental responsivity were tested.

Nadir observations can be calibrated radiometrically

Journal Pre-proof

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Journal Pre-proof

Declaration of interests

The authors declare that they have no known competing financial interests or personal relationships that could have appeared to influence the work reported in this paper.

The authors declare the following financial interests/personal relationships which may be considered as potential competing interests:

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